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A.R C. Technical Report

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Report of the Pressure Panel on
the Definitions of Quantities
having the Dimensions of Pressure

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of Quantities having the Dimensions of Pressure

7th August, 1956

1. Introduction

In a recent **publication**¹ the problem of defining the thrust of a jet **engine**, or the internal drag of a **ducted** body, is **considered** in **some** detail. This Report was compiled by a **special** Panel of the A.R.C. and, during its discussions, concern was expressed at the position regarding the definitions of certain quantities having the dimensions of pressure; **in** some cases **the** most widely-used definitions were not rigorous, in others a **variety** of **names** **were** **in** use. In addition, some quantities of **considerable** value **in** fluid dynamics were either unnamed **or** had been given misleading **names**. This state of affairs was felt to be **undesirable**. As a result, the A.R.C. set up a small Panel to consider the matter more fully. The Panel contained **representatives** from both of **the** two **important** groups **interested** in obtaining **uniformity** in **this** field, those **concerned** with **external** aerodynamics and those primarily interested **in** internal flow problems.

The members of the Panel were:-

Prof. W. A. Mair	Chairman)
Mr. I. M. Davidson	(N.G.T.E.)
Mr. F. B. Greatrex	(S.B.A.C.)
Mr. C. H. E. Warren	(R.A.E.)
Prof. A. D. Young	
Mr. E. W. E. Rogers	(N.P.L., Secretary).

In **the** following paragraphs, the recommendations **agreed** by the Panel **are** set out; for **convenience**, **the** formal definitions of the various **quantities** are **gathered** together **in** an Appendix. **The** **definitions** have been **framed** so that **they** are applicable to the flow of a continuous fluid. For **low-density** (**i.e.**, **free** **molecular**) gases, different definitions would be required.

2. Quantities having the Dimensions of Pressure

The **pressure** at a point within a fluid can be **formally** defined as the mean of the normal components of stress on **three** **mutually-perpendicular** **elements** of **surfaces** at that point, at rest relative to the fluid. For a fluid **in** motion the **pressure** is usually **called** static pressure to distinguish it from other quantities also having the dimensions of pressure, and **the** Panel recommends that **the** **use** of this **term** should **continue**. The static pressure is **the** pressure measured by an **infinitesimal** instrument at **rest** relative to the fluid, or by a correctly designed and aligned static tube. If the fluid is considered as **being** **composed** of discrete **molecules**, **the** static **pressure** can be regarded as arising from the random motion of these **molecules**, a motion which is **superimposed** on the **general** directed motion of **the** fluid.

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If the fluid is brought to rest by a process which is both isentropic and adiabatic[†], the pressure increases to a maximum value. The Panel recommends that this maximum value be called the total pressure of the fluid. Reservoir pressure may be used as an alternative name, where this is more descriptive or is preferred.

In the past, the term 'stagnation pressure' has sometimes been used synonymously with total pressure. The Panel recommends that stagnation pressure be used only to describe the pressure at a stagnation point on a body, defined, for the purposes of this Report, as a point where the velocity of the fluid is zero relative to the body, and where the streamlines divide and continue downstream to follow the contour of the body,

The Panel recommends that the term dynamic pressure be used for the difference between the total and static pressures. Bernoulli's equation shows that for an incompressible fluid this is equal to half the product of the density and the square of the velocity, i.e., the kinetic energy per unit volume of the fluid. For a compressible fluid this equality is no longer valid and the Panel recommends that the kinetic energy per unit volume should be called the kinetic pressure, to distinguish it from the dynamic pressure.

In subsonic flow the total pressure is measured by a pitot tube if this is correctly aligned, provided the Reynolds number of the tube is not very small. When the flow is supersonic, however, a *normal* shock wave exists ahead of the tube end the measured pressure is then the total pressure downstream of the shock wave; this is smaller than the total pressure upstream of the shock wave. The Panel recommends that the pressure measured by a correctly-aligned pitot tube should be called the pitot pressure for both subsonic and supersonic flows.

The difference between the pitot and static pressures is commonly used to determine the speed of an aircraft. When the speed is subsonic the quantity measured is the dynamic pressure, but at supersonic speeds the observed pressure difference is smaller than the dynamic pressure. It may be convenient in certain types of work (e.g., the determination of position error) to have a name for the difference between the pitot and static pressures, and the Panel suggest that indicated dynamic pressure should then be used.

A quantity which often arises when changes of momentum are being considered is the product of the density and the square of the velocity. This quantity is the momentum flux per unit area across a surface normal to the streamlines, and at present has no well-established name. The Panel suggests that it should be called the momentum pressure! if a need is felt for a more explicit form of this name momentum-flux pressure is suggested.

The sum of the static pressure and the momentum pressure arises naturally in applying the momentum equation to determine the force on a body; for this reason the Panel suggests that this quantity be called the reaction pressure.

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[†]It is possible for an irreversible process to maintain constant entropy (i.e., to be isentropic) by transferring heat across the boundaries of the flow. In the process being considered this heat transfer must not be allowed to occur; hence the need for the double condition of isentropic and adiabatic flow.

3. Concluding Remarks

It is hoped that a general adoption of the names and definitions contained in this Report will reduce the present ambiguity in the field of fluid dynamics and will encourage uniformity between the various branches of the subject, each of which has hitherto tended to develop its own nomenclature.

APPENDIX

Definitions of Quantities having the Dimensions of Pressure

The following definitions relate to a continuous fluid.

1. Static pressure.- The mean of the normal components of stress on three mutually-perpendicular elements of surface, at rest relative to the fluid. It is the pressure measured by an infinitesimal instrument at rest relative to the fluid.
2. Total pressure.- The pressure which would arise if the fluid were brought to rest isentropically and adiabatically.
3. Stagnation pressure.- The pressure at a stagnation point on a body.
4. Dynamic pressure.- The difference between the total pressure and the static pressure.
5. Indicated dynamic pressure.- The difference between the pitot pressure and the static pressure.
6. Kinetic pressure.- The kinetic energy per unit volume of the fluid. It is equal to half the product of the fluid density and the square of the velocity.
7. Pitot pressure.- The pressure measured by a pitot tube correctly aligned with the stream. At a point where the flow is subsonic it is equal to the total pressure[†]; at a point where the flow is supersonic it is equal to the total pressure behind a normal shock wave.
8. Momentum pressure.- The momentum flux per unit area across a surface normal to the local flow. It is equal to the product of the fluid density and the square of the velocity.
9. Reaction pressure.- The sum of the static pressure and the momentum pressure.

References/

[†]The exceptional case of very small tube Reynolds number is not considered here.

References

<u>No.</u>	<u>Author(s)</u>	<u>Title, etc.</u>
1		Report of the Definitions Panel on the definitions of the thrust of a jet engine and of the internal drag of a ducted body . C.P. 190. 20th May, 1954. J. R. Ae. Soc., Vol. 59, pp. 517-526. (August, 1955.)

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